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EVALUATION OF A PROPOSED MODIFIED LOG-GAMMA CONFIDENCE BOUND METHOD FOR FLEET MISSILE SYSTEM RELIABILITY

by

Peter Allen Craig

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Thesis Advisor:

M. Woods

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EVALUATION OF A PROPOSED MODIFIED LOG-GAMMA CONFIDENCE BOUND METHOD FOR FLEET MISSILE SYSTEM RELIABILITY

by

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ABSTRACT

A statistical method is evaluated to determine its accuracy for estimating lower confidence bounds on system reliability of a mixture of missile configurations using component data. Monte Carlo simulations are used to establish the accuracy of these bounds.

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I. INTRODUCTION

A statistical method has been proposed which obtains a lower confidence bound on system reliability. It is a modified log-gamma procedure developed to measure fleet missile system reliability. Monte Carlo simulations were performed to evaluate its accuracy as an estimate for system reliability. Five hundred simulations were run for each of twelve cases examined at 80% and 90% confidence levels. The results of these simulations are included in this paper. Additional simulations were performed with minor modifications to the proposed log-gamma method. These changes are documented and the results are included. A comparison was made between the two versions on their accuracy for estimating the lower confidence bound on system reliability.

The reliability equations were applied to a hypothetical fleet missile system configuration and analyzed for changes in test sample sizes. Component reliabilities and weighting factors. The proposed procedure was determined to be significantly inaccurate for small and large amounts of accumulated test data on missile components. It also has the distracting defect that larger lower confidence bounds are obtained from data sets with one failure than those obtained from data sets with zero failures.

II. MODIFIED LOG-GAMMA METHOD

The log-gamma method, in its more general form, can apply to nonseries as well as to fleet-mixture populations. The underlying theory is contained in [Ref. 1]. Examples of cases where it is suspect have been included in the following chapter. The procedure below describes the proposed modified log-gamma method as it is applied to a series system.

Assume that in a series system there are k components each with a sample size n_i , where $i=1,2,\ldots,k$. Let the number of failures be f_i for $i=1,2,\ldots,k$. Consider first the case when there is at least one failure in the system. Thus $\mathbb{E} f_i > 0$.

Let

$$\hat{R}_{i} = 1 - \frac{f_{i}}{n_{i}}$$
 (2.1)

be the point estimates of the i-th component reliability. Then the equation

$$\hat{R} = \frac{k}{\pi} \hat{R}_{i}$$

$$i=1$$
(2.2)

is the point estimate of the system reliability. Define

$$\bar{R} = R^{1/k} \tag{2.3}$$

and

$$\hat{V} = (1 - \bar{R}) \sum_{i=1}^{k} \frac{1}{n_i}$$
 (2.4)

 \hat{V} is used as an estimate of the variance of $-\ln \hat{R}$. It is assumed that the distribution of $-\ln \hat{R}$ can be approximated by a gamma distribution as follows

$$f(z) = \frac{z^{L-1} e^{Lz/\ln R}}{(\frac{-\ln R}{L}) \Gamma(L)}, z \ge 0$$
 (2.5)

where $z = -\ln \hat{R}$ and L and $(\frac{-\ln R}{L})$ are parameters.

Let

$$L^* = \frac{(-\ln \hat{R})^2}{\hat{V}}$$
 (2.6)

and

$$\hat{L} = L^* + 2.25$$
 (2.7)

L* is the method-of-moments estimate of the shape parameter. A constant term 2.25 is added to L*, the shape parameter estimate in the proposed modified log-gamma procedure. The lower $(1-\alpha)$ confidence bound, $\underline{R}(1-\alpha)$ is given by solving the equation

$$\underline{R}(1-\alpha) = \hat{R}^{(2\hat{L}/\chi^2_{2\hat{L}},\alpha)}$$
 (2.8)

where $\chi^2_{2\hat{L},\alpha}$ is the lower α -quantity of the chi-square distribution with $2\hat{L}$ degrees of freedom. Interpolation is required if $2\hat{L}$ is noninteger.

If there are zero failures in the system ($\sum f_i = 0$), let

$$N^* = \frac{k}{\sum_{i=1}^{k} \frac{1}{n_i}}$$
 (2.9)

where N* is defined to be the effective sample size. Then the lower $1-\alpha$ confidence bound $\underline{R}(1-\alpha)$ is computed according to a binomial confidence bound based on zero failures out of N* trials (i.e., $\underline{R}(1-\alpha) = \sqrt[N*]{\alpha}$). If N* is noninteger then linear interpolation is recommended in the proposed procedure but it is not necessary because the same formula could be used for N* an integer.

The modified log-gamma method has been described here for both zero failures and one or more failures in series. The more general form of this method was applied to an actual missile system configuration to determine the lower confidence bounds. The program used to evaluate its accuracy has been included in Appendix B. The complete listing and definitions of the variables used in the program are listed in Appendix A. A description of the more generalized method is described as it was applied to the specific missile system simulated.

In the fleet missile system examined there were different groups of missiles with different configurations. The population was therefore not homogeneous and weights were

assigned to the different groups. There were 14 components in the system modeled and eight mixture weights for the subgroups. The input data consisted of f_i (the number of failures in the i-th component), n_i (the sample size for the i-th component), M_i (the exponent of each component) and C_j (the weights applied to each subgroup). Point estimates for this system were defined as follows

$$\hat{R}_{i} = 1 - \frac{f_{i}}{n_{i}}$$
 (2.10)

$$\hat{p}_{R} = \int_{i=1}^{5} R_{i}^{M_{i}}$$
 (2.11)

$$\hat{p}_{N} = \prod_{i=6}^{10} R_{i}^{M_{i}}$$
 (2.12)

with the subgroup reliability point estimates being

$$\hat{R}^{(1)} = \hat{p}_{R} \hat{R}_{11} \hat{R}_{12}$$

$$\hat{R}^{(2)} = \hat{p}_{N} \hat{R}_{11} \hat{R}_{12}$$

$$\hat{R}^{(3)} = \hat{p}_{R} \hat{R}_{13} \hat{R}_{14}$$

$$\hat{R}^{(4)} = \hat{p}_{N} \hat{R}_{13} \hat{R}_{14}$$

$$\hat{R}^{(5)} = \hat{p}_{R} \hat{R}_{13} \hat{R}_{12}$$

$$\hat{R}^{(6)} = \hat{p}_{N} \hat{R}_{13} \hat{R}_{12}$$

$$(2.13)$$

$$\hat{R}^{(7)} = \hat{p}_R \hat{R}_{11} \hat{R}_{14}$$

$$\hat{R}^{(8)} = \hat{p}_N \hat{R}_{11} \hat{R}_{14}$$

and

$$\hat{R} = \sum_{j=1}^{8} c_j \hat{R}^{(j)}$$
 (2.14)

The variance of -ln \hat{R} is then estimated by \hat{V} given by equation (2.15)

$$\hat{V} = \frac{1}{\hat{R}^2} \sum_{i=1}^{8} \sum_{j=1}^{8} c_i c_j \hat{R}^{(i)} \hat{R}^{(j)} S_{ij}$$
 (2.15)

where S_{ij} estimates the $cov(z^{(i)}, z^{(j)})$ and where $z^{(i)} = -\ln \hat{R}^{(i)}$. The estimates S_{ij} are found by solving the following equations.

$$z_{i} = -\ln \hat{R}_{i} \qquad (2.16)$$

$$\overline{R} = \exp(-\sum_{i=1}^{14} M_i z_i / \sum_{i=1}^{14} M_i)$$
 (2.17)

and

$$V_R = (1 - \overline{R}) \sum_{i=1}^{5} \frac{M_i^2}{n_i}$$
 (2.18)

$$V_{N} = (1 - \overline{R}) \sum_{i=6}^{10} \frac{M_{i}^{2}}{n_{i}}$$
 (2.19)

$$V_i = (1 - \overline{R})/n_i$$
 , $i = 11,...,14$ (2.20)

Then the S_{ij}'s are solved by the equations listed in the program in Appendix B and repeated below.

Finally,

$$\hat{L} = \frac{(-\ln \hat{R})^2}{\hat{V}} + 2.25 \qquad (2.22)$$

and DF, the degrees of freedom, is equal to

$$DF = 2\hat{L}$$
 (2.23)

Thus

$$\underline{R}(1-\alpha) = \hat{R}^{(DF/\chi_{DF,\alpha}^2)}$$
 (2.24)

III. EVALUATION PROCEDURE

The equation for system reliability is

$$R_{s} = \sum_{j=1}^{L} w_{j} \prod_{i=1}^{k} p_{i}^{M_{i}}$$

$$(3.1)$$

where

L = number of subsystems

 w_i = the weighting factor of the j-th subsystem

k = the number of components

p; = the reliability of the i-th component

M; = the exponent of the i-th component

The computer program modeled a system that had 8 subsystems and 14 components. System reliability (RS) was determined for each case and a lower confidence bound for $\alpha=.1$ and $\alpha=.2$ was computed. Random numbers were drawn from a shuffled random number generator [Ref. 3]. Inverse chi-square values were determined using the international mathematical and statistical library (IMSL) routine called MDCHI. All computations were done in single precision arithmetic, coded in FORTRAN, using an IBM 360 computer.

A. ZERO FAILURE VS ONE FAILURE CASE

An examination of two cases revealed a shortcoming and a motivation for evaluating the modified log-gamma procedure. These two examples are considered below.

Example 1.

Let k, the number of components in the system, be 14 and let R_i, the component reliabilities, all equal .99. The sample sizes (mission trials) and failures for each component are listed in Table I. The lower 90% confidence limit on system reliability is desired.

Table I

Component

1 2 3 4 5 6 7 8 9 10 11 12 13 14

Mi: # mission trials 10 10 10 10 500 10 10 10 500 10 10 10 10

fi: # failures 0 0 0 0 0 0 0 0 0 0 0 0 0 0

When the $\Sigma f_i = 0$ the modified log-gamma procedure defines N*, the effective sample size, as

$$N^* = \frac{k}{\sum_{i=1}^{k} \frac{1}{n_i}}$$
 (3.2)

For the data given in the table above N* is equal to 11.628. For this procedure the lower 1- α confidence bound $\underline{R}(1-\alpha)$ is computed according to a binomial confidence bound based on zero failures out of N* trials. The value obtained for $\underline{R}(1-\alpha)$ was .820.

Example 2.

Let sample sizes and f_i (failures for each component) be given in Table II. Again the lower 90% confidence limit on system reliability is desired.

Table II

								∞	mpor	ent					
		1	2	3	4	5	6	7	8	9	10	11	12	13	14
M _i :	# mission trials	10	10	10	10	500	10	10	10	10	500	10	10	10	10
f _i :	* failures	0	0	0	0	1	0	0	0	0	0	0	0	0	0

When $\Sigma f_i \neq 0$ the modified log-gamma method solves for $\underline{R}(1-\alpha)$ (the lower confidence bound) by the following procedure. Let

$$\hat{R}_{i} = 1 - \frac{f_{i}}{n_{i}}$$
,
 $\hat{R}_{5} = .998$, (3.3)
 $\hat{R}_{i} = 1$ for $i \neq 5$

be the point estimate of the i-th component reliability. Then

$$\hat{R} = \prod_{i=1}^{k} R_i = .998$$
 (3.4)

is the point estimate for system reliability. Define

$$\bar{R} = \hat{R}^{1/k} = .99986$$
 (3.5)

$$\hat{V} = (1-\bar{R}) \sum_{i=1}^{k} \frac{1}{n_i} = .000172$$
 (3.6)

where \hat{V} estimates the variance of $-\ln \hat{R}$.

Let

$$L^* = \frac{(-\ln \hat{R})^2}{\hat{V}} = .02328 \qquad (3.7)$$

and define

$$\hat{L} = L^* + 2.25 = 2.27328$$
 (3.8)

where 2.25 is the correction term and L* is the method-of-moments estimate of the shape parameter. Then the lower $1-\alpha$ confidence bound, $R(1-\alpha)$ is computed by solving

$$\underline{R}(1-\alpha) = \hat{R}^{(2L/\chi_{2L}^2,\alpha)}$$
 (3.9)

where $\chi^2_{2\hat{L},\alpha}$ is the lower α quantity of the chi-square distribution with $2\hat{L}$ degrees of freedom. In example 2 $\underline{R}(1-\alpha)$ is equal to .993

These two examples have shown the shortcoming of this method. The lower confidence bound for one failures is higher than the lower confidence bound for zero failures.

B. SIMULATION RESULTS

The lower confidence bound values obtained for the twelve cases studied have been listed in Table III. RS is the system reliability, ACV is the actual confidence value computed by the modified log-gamma method and $R(1-\alpha)*500$ is the percentile value of the 500 ordered $R(1-\alpha)$ estimates for $\alpha = .1$ and $\alpha = .2$. N(I), RI(I) and W(I) are the respective sample sizes, reliabilities and weights assigned to each case.

For example, in case number 3 the number of components k, is equal to 14 with the sample sizes equal to 50. For $i \neq 5$ or 10 and 250 for i = 5 or 10. The reliabilities of each component is .99 and the 8 weights are all equal to .125. System reliability, R_S , was computed to be .816 and for $\alpha = .1$ the 450-th value in the ordered 500 LCL estimates was .895. The R_S value of .816 was the 35th of the 500 ordered LCL estimates yielding an actual confidence level of 7.8%. Likewise for $\alpha = .2$ the 400-th value in the ordered 500 LCL estimates was .898. The R_S value of .816 was the 13-th of the 500 ordered LCL estimates yielding an actual confidence level of 2.8%. In only one case (case 8) did the actual confidence value approach that of the system reliability as a lower bound.

An examination of the MLG (modified log-gamma) procedure questioned the inclusion of the correction term 2.25. Additional simulations were run on the same twelve cases when this correction term was removed and the degrees of freedom bounded

below by 1.0. The results obtained from this modification, while an improvement, were still far from providing accurate lower bounds on the system. The values determined from these runsaire listed in Table IV. ACV values of 100% indicate that the system reliability was greater than all 500 estimates.

It would appear that in order to generate more estimates less than RS the exponent, $2L/\chi^2_{2L,\alpha}$, needs to take on larger values. Adding a constant term such as 2.25 yields more values for $\underline{R}(1-\alpha)$ that are greater than RS. Indeed, Tables III and IV did show this to be the case. As the exponent becomes larger (the chi-squared value smaller) the confidence level decreases. The estimate for \hat{L} used in generating the values listed in Table IV seem more accurate when used in the modified log-gamma procedure.

This modification still left much room for improvement. A closer reivew of the MLG method pointed to the estimate of the shape parameter as a possible cause of the extreme results. Since $Z = -\ln \hat{R}$ its distribution was approximated by a two-parameter gamma distribution. Then

$$f(z) = \frac{z^{L-1} e^{Lz/\ln R}}{(-\frac{\ln R}{L})^{L} \Gamma(L)}, z \ge 0$$
 (3.10)

where L and $\left(-\frac{\ln R}{L}\right)$ are the parameters. Then

$$E(z) = L \cdot \frac{(-\ln R)}{L} = -\ln R$$
 (3.11)

and

$$Var(z) = L(\frac{-\ln R}{L})^2 = \frac{\ln^2 R}{L}$$
 (3.12)

Note:

$$L = \frac{\ln^2 R}{Var(z)} = \frac{[E(z)]^2}{Var(z)}$$
 (3.13)

The proposed estimator \hat{L} for L is

$$\hat{L}_1 = \frac{z^2}{\text{Var}(z)} \tag{3.14}$$

Since $L = \frac{[E(z)]^2}{Var(z)}$ it would appear that $\hat{L} = \frac{[E(z)]^2}{Var(z)}$ would

be a better estimator for L. Since

$$[E(z)]^2 = E(z^2) - Var(z)$$

we have

$$\hat{L} = \frac{E(z^2) - Var(z)}{Vaf(z)}$$
 (3.15)

and since z^2 is unbiased for $E(z^2)$ we get

$$\dot{L} = \frac{z^2 - var(z)}{var(z)} = \frac{z^2}{var(z)} - 1$$
 (3.16)

Note that this is a departure from \hat{L}_1 in the proposed method. Thus the shape parameter L can be estimated by Eq. 3.16 above. This estimate is different from the original version of the MLG method.

Substituting this new value for L and bounding the degrees of freedom by 1.0, so as not to obtain a negative value, the results show a little more improvement. The results obtained from this second modification are listed in Table V.

TABLE III

Accuracy of $\underline{R}(1-a)$ as a 100(1-a)% Lower Confidence Limit for RS (Correction Term Equal to 2.25)

CA SE	-	N(I),RI(I) AND W(J)	R.S	ALPHA	ALPHA (1-4)*500	ACV	STANDARD DEVIATION OF R (1-a)
-	141	EXCEPT N(5) = 50, N(10) = 50	.816		.872	66.42	.062
		RI(I)=.99, I=1,2,,14 W(J)=.125, J=1,2,,8					
1	14	14 N(I)= 20 I= 1 2	. 616		900	40.2%	
		RI(1)=.99, I=1,2,,14 W(J)=.125, J=1,2,,8					
7	41	14 N(1)=50 I= 1, 2, 14 EXCEPT A(5)=250 AND N(10)=250	. 816	.1	898	7.8%	. 629
		RI(I)=.99, I=1,2,,14 W(J)=.125, J=1,2,,8	,				
*	14	14 N(I)= 10, I= 1, 2, (I 0)=80	. 816	.2	903	43.4%	. 067
		RI(1)=.59, I=1,2,,14 W(J)=.125, J=1,2,,8					

TABLE III (Continued)

CEVIATION OF R (1-a)	044		.026			040.			
A C V	30.6%		5.6%			63.0% 82.8%			
ALPHA (1-a) \$500	006:		.898			.878			
ALPHA	.2		.2			.1			
RS	. 816		. 816			. 863			
K N(I),RI(I) AND W(J)	14 N(I)=20 I= 1 2	RI(11) =, 99, 1=1,2,,14	14 N(I)=50 L = 1 2 3 14 EXCEPT K(5) = 400 AND N(10) = 400	RI(1) =. 99, I=1,2,,14	W(J)=.125, J=1,2,,8	14 N(1): 5 4 99 1995	RI(I): 98 . 957 . 558 . 99	RICI): . \$55 . \$58 . \$98 . 59 N. []: . \$55 . \$58 . \$98 . 99 RICI]: . \$93 . 98 . 99	W(J): .0625 .125 .125 .125
	1			_	_				
CASE NO.	2		9			1			

TABLE III (Continued)

	N(I),RI(I) AND W(J)		AL PHA		ACV	DEVIATION OF
	RI(1): 15 5C 20 30	.821	.1	.886	38.42	049
	RI(1): .598 .99 .993 .99					
T.	RI[1]: 597 .953 .97 .99					
G.	RI(I): 8 797					
-	W(J)=.125, J=1,2,,8					
· ·	RI(1): 15 5C 20 30	. 665	.2	. 187	23.4%	. 057
CT.	RI(1): 100 5 20 10					
CY	N(1): 20 100 15 30					
CZ.	RI(1): 8 7					
-5	W(J)=.125, J=1,2,,8					
· ·	RI(11): 15 50 20 RI(11): 9975 .995 .9985	. 907	.2	. 936	61.68	.031
C	RICI): 39 . 995 . 995 . 9965					
a	N(I): 10 20 100					
O.	RI(I): 15 30 8 7					
-3	W(J)=.125, J=1,2,,8					

TABLE IV

Accuracy of $\mathbb{R}(1-\alpha)$ as a $100(1-\alpha)$ % Lower Confidence Limit for RS (Correction term equal to 0.0, DF > 1.0)

AL PHA (11-a) #500 ACV DEVIATION OF	.816 100. 2 .146 .859 69.42 .074	.854 69.82 .045		.894 10.0%		.818 86.4% .157 .857 59.4% .074	
RS AL PHA (1-	.816			. 616		. 816	
N(I),RI(I) AND W(J)		14 N(1)=20 I=125, J=1,2,,8 EXCEPT N(5)=100 AND N(10)=100	RI(1)=.99, I=1,2,,14 W(J)=.125, J=1,2,,8	14 M(1)=50, I=1,2,, 14 EXCEPT N(5)=250 AND N(10)=250	RI(I)=.99, I=1,2,,14 W(J)=.125, J=1,2,,8		RI(1)=.99, [=1,2,,14
×	14	141		14		14	
CASE NO.]-	7		,		4	

TABLE IV (Continued)

!									
STANDARD DEVIATION OF R (1-0)	. 051		.025			. 241			
ACV	54.8% 19.6%		1.2%			100.2			
ALPHA (1-a) *500	.882		. 888 . 894			.775			
AL PHA	-1		.2			.2			
» s	. 816		. 816			. 883			
N(I), RI(I) AND W(J)	EXCEPT N(5) = 1 2 14 EXCEPT N(5) = 160 AND N(10) = 160	RI(I)=.99, I=1,2,,14 W(J)=.125, J=1,2,,8	14 N(1)=50 L= 1, 2, 3, 14 EXCEPT A(5)=400 AND	RI(I)=. 99, I=1,2,,14	W(J)=.125, J=1,2,,8	RI(I): 598 .99 .995	RI(I): 9 20 5 4	RI(I): \$95 . \$58 . \$98 . \$9 N(I): \$47 . 7 RI(I): \$93 . 98 . 99	W(J): .0625 .125 .125 .125 .125 .125
¥	14		14			14			
CASE NO.	2		9						

TABLE IV (Continued)

1										
STANDARD DEVIATION OF R (1-0)	.146					022				
A C V	61.82					43.0%				
(1-a)*500	935:					.917				
ALP HA	.1.					.2				
S.	. 883					.883				
N(I), RI(I) AND W(J)	14 N(1): 20 16 28 36 RI(1): .998 .998 .995 .98	RI(II): 997 .998 .99 .995	RI(II): 998 .998 .99 .993	RI(1): 28 36	W(J): 0625 .125 .125 .125	14 N(1): 50 40 70 90	RI(1): 597 .958 .59 .995	RI(1): 50 200 50 40	RI(1): 70 90	W(J): .0625 .125 .125 .125 .125 .125
-	1-					<u> </u>				
CASE NO.	30					5				

TABLE IV (Continued)

ST ANCAR C DEVIATION OF (1-a)	: 675					. 062					245				
AC V	74.23					11.02					55.8% 80.6%				
(1-a) *500	-860					.759					. E E E				
AL PHA	.2					.1					.1				
R S	.821					. 669					205				
NCIDARICID AND WCLD		RI(1): .998 .99 .993 .99	RI(1): .997 .993 .57 .99	RI(1): 98 .97	W(J)=.125, J=1,2,,8		RI(11): 988 .98 .983 .98	RICID: 587 .983 .96 .98	RI(I): 97 .96	W(J)=.125, J=1,2,,8		RI(1): 39 . 959 . 955 . 9965	RILLI: 555 . 965 . 9965	A [(1): 585 .955 .59 . 985	W(J)=.125, J=1,2,,8
×	14					14				_	14				_
CA SE	107					=_					12				

TABLE V

Accuracy OF $\underline{R}(1-a)$ as a 100(1-a)% Lower Confidence Limit for RS (Correction term equal to -1.0; DF \geq 1.0)

CASEI NC.	*	N(I), RI(I) ANG W(J)	RS.	ALP HA	ALPHA (1- 2) \$500	ACV	STANDARD DEVIATION OF R (1-a)	
11-	14	14 N(1)=10, I=1,2,,14	. 616		. 816	100.	.242	11
		EXCEPT N(5)=50, N(10)=50 RI(1)=.99, I=1,2,,14		?	.152	88.8%	• 155	
	_	W(J)=.125, J=1,2,,8						
2	14	14 N(I) =20, I=1 26 and Exc EPT N(5) = 160 and N(10) = 100	.816	.2	.833	86.62	.078	
		RI(I)=.59, I=1,2,,14 W(J)=.125, J=1,2,,8						
3	14	14 N(1) =50, I=1 2 0 14 EXC EPT N(5) = 250 AND N(10) = 250	.816	177	.891	11.82	.027	1
		RI(I)=.99, I=1,2,,14 W(J)=.125, J=1,2,,8						
4	14	EXCEPT N(5)=80, N(10)=80	.816		.700	94.62	. 147	ī
		RI(I) = . 99, I = 1,2,,14 W(J) = .125, J=1,2,,8						

TABLE V (Continued)

	1								
ACV DEVIATION OF	137		. C24			. 30 3			
ACV	74.43		7.8%			100.3			
AL PHA (1-4) * 500	.848		. 894			. 695 . 775			
AL PHA	-5-		-2.			.2			
R.S	.816		.816			.883			
N(1),RI(1) AND W(J)	(N(I)=20, I=1 20. 14 EXCEPT N(5)=160 AND N(10)=160	RI(1)=.99, I=1,2,,14 W(J)=.125, J=1,2,,8	14 N(1)=50, 1=1,2,,14 EXCEPT N(5)=400 AND A(10)=400	RI(1)=.99, I=1,2,,14	M(J)=.125, J=1,2,,8	Z.Z.	RI(1): 5 20 5 4	RI(1): .995 .998 .998 .99 N(1): .4 .7 .7 .99 (RI(1): .593 .96 .99	W(J): 0625 .125 .125 .125 .125 .125
¥	14		7 _			*			
CASE NO.	5		٥			~			

TABLE V (Continued)

STANDARD DEVIATION OF R (1-a)	.134					. 052				
AC v	97.82					62.0%				
R OF (1 -a) *500	.866					908				
AL PHA	.1					.1				
R S	.883					. 883				
פארט)	28 36	. \$9 . 995 . \$9 . 995	20 16 . 593		125 :125	96.	266. 66. 99.	50 40		.125 .125
I),RI(I) AND W(J)	20 16 . 55 .	802 . 765	36 80 598 . 958	28 36 98 .99	.0625 .125	50 46	997 .998	90 200	96 01 58 85	0625 .125
N(I)	RICID:	A ICID:	RICID:	RICED:	00 : (F)M	RICID:	RILLS:	RICID:	RICID:	90 -: (F)M
*	14					141				
CASE NO.	100	_				6				

TABLE V (Continued)

STANDARD DEVIATION OF R (1-a)	. 179					.076					. 163				
A CV	EC.0%					56.62					100.1				
(1-a)*500	. 854					.744					.854 .856				
AL PHA	.1					.2					.2				
S.	.821					699-					106.				
N(I), RI(I) AND W(J)	RICID: 15 50 20 30	RI(I): 998 . 99 . 953 . 99	RICII: \$97 . 953 . 57 . 99	RI(1): 8 797	W(J)=.125, J=1,2,,8	RICIJ: 15 50 20 30	P I(1): 588 .96 .983 .98	RI(I): 987 .983 .56 .98	RI(1): . 67 96	W(J)=.125, J=1,2,,8		RI(1): 39 .999 .995 .9965	RICI1: 10 20 100	RI(1): 15 30 8 7	W(J)=.125, J=1,2,,8
	14					14					14				
CASE NO.	100					=					12				

IV. CONCLUSIONS

Additional simulations on many more cases would be required to determine the particular conditions under which this modified log-gamma method is reasonably accurate. For the cases examined here the proposed procedure remains suspect in estimating lower confidence bounds on system reliability.

APPENDIX A

AA	CORRECTION TERM EQUAL TO 2.25 IN THE MODIFIED LOG-
48	VARIABLE THAT STORES THE CIFFERENCE BETWEEN RS (SYSTEM RELIABILITY) AND RR(400) THE 80TH PERCENT ILE POINT WHEN ALPHA=0.2
ABS	AR SOLUTE VALUE
AC	VARIABLE THAT STORES THE CIFFERENCE BETWEEN RS (SYSTEM RELIABILITY) AND R(450) THE 90TH PERCENT ILE POINT WHEN ALPHA=0.1
ALOG	NATURAL LOGARITHM SUBROUTINE
ALPHA	VARIABLE ASSIGNED A VALUE OF 0.1
ALPHAA	VARIABLE ASSIGNED A VALUE OF 0.2
Δ.Α	ARRAY THAT STORES THE EXPONENTS M SUB I
BLHAT	VARIABLE THAT STORES THE L HAT VALUE
CA	ACTUAL CONFIDENCE LEVEL FOR ALPHA=0.1
CALL	FORTRAN CODE FOR ACCESSING SUBROUTINES
C9	ACTUAL CONFIDENCE LEVEL FOR ALPHA=0.2
CONTINUE	FORTRAN CODE TO CLOSE EACH DO LCCP
EA	CUMMY VARIABLE USED TO DETERMINE THE ACTUAL CONFIDENCE LEVEL
DDF	CEGREES OF FREEDOM
DIMENSION	FORTRAN CODE REQUIRED FOR CIMENSIONING ARRAYS
30	FORTRAN CODE USED TO BEGIN LOOPS
DUM	CUMMY VAR IABLE
EA	DUMMY VARIABLE USED TO DETERMINE ACTUAL CONFIDENCE VALUE
EFFN	EFFECTIVE SAMPLE SIZE
END	FORTRAN CODE REQUIRED TO END PROGRAM
EXP	EXPONENT IAL SUBROUT IN E
FA	DUMMY VARIABLE USED TO DETERMINE THE ACTUAL CONFICENCE VALUE
FLOAT	FORTRAN CCDE USED TO CHANGE INTEGERS TO DECIMAL VALUES
FORMAT	FORTRAN STATEMENT
GA	DUMMY VARA
GS	CONFIDENCE VALUE
GO	FORTRAN CODE USED IN THE -GO TOSTATEMENT
HI STG	SUBROUTINE WHICH GENERATES A HISTOGRAM OF THE

I INDEX VARIABLE IER ERROR VARIABLE IN SUBROUTINE MOCHI VARIABLE USED TO STORE THE NUMBER OF FAILURES (ALSO PART OF THE FORTRAN -- IF-- STATEMENT) IF INDEX VAR IABLE 11 VARIABLE THAT STORES THE INITIAL VALUE FOR CALLING RANDOM NUMBERS ISEED INDEX VARIABLE J VARIABLE THAT STORES THE NUMBER OF FAILURES PER COMPONENT JF INCEX VAR IABLE JJ INDEX VAR IA3 LE JM VARIABLE THAT STORES THE NUMBER OF COMPONENTS (ALSO USED AS AN INDEXING VARIABLE) KJ INDEX VARIABLE VARIABLE THAT STORES THE NUMBER OF SUBSYSTEMS (ALSO USEC AS AN INCEXING VARIABLE) COUNTER VARIABLE MC MDCHI INVERSE CHI SQUARE SUBROUTINE MM INDEX VARIABLE ARRAY THAT STORES THE K SAMPLE SIZES N NC COUNTER VARIABLE VARIABLE THAT STORES THE NUMBER OF CASES NCASE NN INDEX VARIABLE CVFLOW SUBROUTINE REQUIRED FOR RANDOM NUMBER GENERATION P ARRAY THAT STORES THE UNIFORM RANDOM NUMBERS PD VARIABLE THAT STORES THE ALPHA VALUE OF .1 PN VARIABLE THAT STORES A POINT FR VARIABLE THAT STORES A POINT ESTIMATE R ARRAY THAT STORES THE LOWER CONFIDENCE BOUND VALUE WHEN ALPHA=0.1 VARIABLE THAT STORES THE R(450) VALUE 99 RBAR VARIABLE THAT STORES R BAR RBN NUMBER OF REENTRY ECDIES PER MISSILE RCEN CUMMY VARIABLE USED TO COMPUTE RBAR READ FORTRAN STATEMENT VARIABLE THAT STORES THE INVERSE OF THE EFFECTIVE SAMPLE SIZE REFFN VARIABLE THAT STORES THE SUM OF THE WEIGHTED SUBGROUP RELIABILITY ESTIMATES

RHAT

RI	ARRAY THAT STORES THE INPUTED RELIABILITY VALUES
RIHAT	ARRAY THAT STORES THE COMPUTED RELIABILITY VALUES
RMEAN	VARIABLE THAT STORES THE MEAN OF THE R ARRAY
RNUM	DUMMY VARIABLE USED TO COMPUTE RBAR
RR	ARRAY THAT STORES THE LOWER CONFIDENCE EDUNCS WHEN ALPHA=0.2
RRB	VARIABLE THAT STORES THE RR (400) VALUE
RRMEAN	VARIABLE THAT STORES THE MEAN OF THE RR A FRAY
FRVAR	VARIABLE THAT STORES THE VARIANCE OF THE RR ARRAY
25	VARIABLE THAT STORES THE TOTAL SYSTEM RELIABILITY
RUHAT	ARRAY THAT STORES THE SUBGROUF FELIABILITY ESTIMATES
RVAR	VARIABLE THAT STORES THE VARIANCE OF THE F ARRAY
S	ARRAY THAT STORES THE VAR/COV MATRIX
SCR	VARIABLE THAT STORES THE STANDARD DEVIATION OF THE R ARRAY
SORR	VARIABLE THAT STORES THE STANDARD DEVIATION OF THE RR ARRAY
SCRT	SUBPOUT INE THAT SOLVES SQUARE ROOTS
SRAND	SUBROUTINE THAT IS THE SHUFFLED RANDOM NUMBER GENERATOR
STOP	FORTRAN REQUIRED CODE
SUM	DUMMY VARIABLE USED THROUGHOUT THE PROGRAM
TO	PART OF THE FORTRAN GO TO STATEMENT
٧	ARRAY THAT STORES 4 VARIANCE ESTIMATES FOR COMPONENTS 11 THROUGH 14
VFAT	VARIABLE THAT STORES THE VARIANCE ESTIMATE FOR -LN(RHAT)
VN	VARIABLE THAT STORES THE VARIANCE ESTIMATE FOR -LN(PN)
VR	VARIABLE THAT STORES THE VARIANCE ESTIMATE FOR -LN(PR)
VX	DUMMY VARIABLE USED IN THE MOCHI SUBROUTIME
VY	DUMMY VARIABLE USED IN THE MOCHI SUBROUTINE
•	ARRAY THAT STORES THE WEIGHTED VALUES OF EACH SUBSYSTEM
WRITE	FORTRAN STATEMENT
2	CUMMY VARIABLE USED TO DETERMINE RBAR

AFPENCIX B

```
DIMENSION R(5C0), RI(50), W(50), N(50), RIHAT(50), AM(50), AW(50), V(4), S(8, 8), RR(500), P(5CC)

CALL CVFLCW

NCASE = 0

ISEED = 134869
COCOCO COCOCO COCOCO
        REALING IN THE INPUT PARAMETERS K, L, RBN, A, ALPHA AND ALPHAA
        READ (5,330) K, L, RBN, AA, ALPHA, ALPHA
        REACING IN
COMPONENT
                        N(I) --- THE NUMBER OF RANCOM NUMBERS PER
    20 REAC (5,340) (N(I), I=1,K)
        READING IN THE COMPONENT/FUNCTION RELIABILITIES
        REAC (5,350) (RI(I), I=1,K)
NCASE = NCASE+1
IF (ISEED.GT.134869) GO TO 30
00000 000000 000000 00000
         READING IN THE EXPONENTS M SUB I
        REAC (5,360) (AM(I), I=1,K)
        REACING IN THE WEIGHTS FOR EACH SUBSYSTEM/GROLP
    30 REAC (5,370) (W(I), I=1,L)
         STARTING THE MAIN LCCP FOR 500 SIMULATIONS
        00 190 7 = 1,500
        LOCPING FOR EACH COMPONENT AND DRAWING THE RANDOM NUMBERS
         IF = C
         00 50 J=1 ,K
         CALL SRANC (ISEED, P, JJ)
C
    CO 40 JM=1, JJ
IF (P(JM).GT.RI(J)) JF=JF+1
40 CONTINUE
C
         IF = IF+JF
         VARIABLE "IF" COUNTS THE FAILURES
        RIHAT(J) = 1.-(FLOAT(JF)/FLOAT(JJ))
CONTINUE
```

```
IF (IF. EQ.O) GO TO 170
COC
            COMPUTING THE POINT ESTIMATES
            PN = 1.
PR = 1.
           CO 60 J=1,5
FR = (RIHAT(J)**AM(J))*FR
PN = (RIHAT(J+5)**AM(J+5))*PN
            CONT INUE
COCOCO
            COMPLTING THE SUBGROUP RELIABILITY ESTIMATES
                                PR*RI(11)*RI(12)
PN*RI(11)*RI(12)
PR*RI(13)*RI(14)
PN*RI(13)*RI(14)
PR *R I(13)*RI(12)
PN*RI(13)*RI(12)
PR *R I(11)*RI(14)
PN *R I(11)*RI(14)
           RUHAT(1) =
RUHAT(2) =
RUHAT(3) =
RUHAT(4) =
RUHAT(5) =
RUHAT(6) =
RUHAT(7) =
RUHAT(8) =
CCC
            COMPUTING RHAT AND CALLING IT BY THE SAME NAME--RHAT
            RHAT = 0.
C
      DC 7C J=1,L
RHAT = (W(J) *RUHAT(J))+RHAT
70 CONTINUE
COCOCOCO
            ESTIMATING THE VARIANCE OF -LN(RHAT)-----VHAT
            STEP 1:
                           DETERMINING RBAR
            RNUM = 0.
RDEN = 0.
           C3 80 J=1,K
RNUM = (-ALGG(RIHAT(J))*AM(J))+RNUM
RDEN = AM(J)+RDEN
CONTINUE
      Z = (-RNUM)/RDEN

IF (Z.LT.0) GC TO 90

RBAR = EXP(Z)

GD TO 100

90 RBAR = 1./EXP(ABS(Z))
            STEP 2:
                              DETERMINING THE VARIANCE ESTIMATES
            VR = C.
VN = O.
    100
C
            CO 110 J=1, 5

VR = (AM(J)**2/FLOAT(N(J)))+VR

VN = (AM(J+5) **2/FLOAT(N(J+5)))+VN

CONTINUE
            VR = (1.-RBAR)*VR
VN = (1.-RBAR)*VN
C
            DO 120 J=1,4
V(J) = (1.-RBAR)/FLCAT(N(J+10))
```

```
120 CONTINUE
000000
            STEF 3:
                              FINAL SOLUTIONS FOR COVARIANCE ESTIMATES
                1234567811111111222223333344445555667
            1934567811111111199999999999944445556667
10069966996699669666996669666969696969
                             VR
0.
VR+V(2)
V(2)
VR+V(1)
V(1)
0.
VN
                         •
                             VN
V(2)
VN+V(2)
V(1)
VN+V(1)
V(3)+V(4)
V(3)
V(3)
V(3)
V(4)
V(4)
                         . . . .
                          =
                             V(4)
V(3)
V(4)
V(4)
V(4)
V(4)
V(4)
V(3)+V(2)
VR
                          = = =
                         =
                         . . .
                             0.
VN
V(1)+V(4)
COCO
            FILLING IN THE REST OF THE VAR/COVAR MATRIX
            DU 140 MM=1 , L
C
    C3 130 NN=1,L
S(NN,MM) = S(MM,NN)
130 C3NTINUE
    140 CONTINUE
000000
           SOLVING THE OVERALL EQUATION FOR WHAT
            VHAT = 0 .
C
            CJ 160 J=1.L
C
           DJ 150 KJ=1,L
VHAT = W(J) *W(KJ) *R UHAT(J) *RUHAT(KJ) *S(J,KJ) + \HAT
CONTINUE
C
    160 CONTINUE
C
            VHAT = VHAT ( RHAT + 2)
```

C

```
CCCC
           COMPUTING L HAT
           ELFAT = ( (-ALOG(RHAT)) **2/VHAT) - 1.0
0000000
          COMPUTING THE DEGREES OF FREEDOM-DDF AND SOLVING FOR R OF (1-ALPHA) WHEN THE SUM OF THE FAILURES DOES NOT EQUAL ZERO
           CDF = 2.*BLHAT
IF (CGF.LT.1.0) DDF=1.0
PD = .1
CALL MDCHI (PC.DDF, VX, IER)
          FD = .2

CALL MDCHI (PD,ODF, VY, IER)

R(I) = R FAT ** ((DDF)/VX)

RR(I) = RHAT** ((DDF)/VY)

GO TO 190
COCOCO
           COMPUTING RELIABILITY ESTIMATES WHEN THE SUM OF THE FAILURES IS GREATER THAN ZERO
   COC
           CALL HISTG (R.500.0)
CALL HISTG (RF,500.0)
COCOCO
           COMPUTING THE TOTAL SYSTEM RELIABILITY
           DUM = 0.
C
           DC 210 J=1,L
SUM = 1.
          D3 200 I=1,K
SUM = SUM*(RI(I)**AM(I))
CONT INU E
   200
   DUM = (W(J) *SLM)+DUM
210 CONTINUE
C
           RS = DUM
C
           AC = RS-R(450)
AB = RS-RR(400)
R9 = R(450)
FRB = RR(400)
           COMPUTING THE SAMPLE VARIANCE FOR ALPHA=. 1-- SCR AND ALPHA=.2---SORR
```

```
SUM = 0.
C
                 D7 220 I=1,500
SUM = R(I)+SUM
DUM = RR(I)+DLM
CONTINUE
      220
C
                 RMEAN = SUM/500.
RRMEAN = CUM/500.
SUM = 0.
DJM = 0.
C
                 DJ 230 I=1,500
SUM = (R(I)-RMEAN) **2+SUM
LUM = (RR(I)-RRMEAN) **2+DUM
      230 CONTINUE
                 FVAF = SUM/499.
RR VAR = DUM/499.
SDR = SQRT(RVAR)
SDRR = SQRT(RFVAR)
COCOO
                 DETERMINING THE ACTUAL CONFIDENCE VALUE
                  MC = 0
C
                 CC 250 I=1,500
IF (RS.GT.R(I)) GO TC 240
IF (RS.GT.R(I)) GO TC 240
DA = R(I)-RS
EA = RS-R(MC)
IF (DA.LT.EA) GO TO 260
CA = FLOAT(MC)/500.
GJ TO 270
MC = MC+1
CONTINUE
      240
250
C
      260 CA = FLOAT(I)/500.
270 CA = CA*100.
CCC
                 NC = 0
C
     D) 290 I=1,5CC

IF (RS.GT.RR(I)) GD TD 280

FA = RR(I)-RS

GA = RS-RR(NC)

IF (FA.LT.GA) GD TD 300

C3 = FLOAT(NC)/50D.

GD TD 310

280 NC = NC+1

290 CJNTINUE
C
      300 CB = FLCAT(I)/500.
310 CB = CB*1CO.
00000
                 PRINTING THE FINAL RESULTS FOR EACH CASE
                WRITE (6,380) NCASE

WRITE (6,390)

WRITE (6,400) RS,RB,AC,SDR,CA

WRITE (6,410)

WRITE (6,420) RS,RRB,AB,SDRR,CE

ISEED = ISEED+27

IF (NCASE.EG.8) GO TO 320

STOP
      320
```

```
330 FJRMAT (12,12,F5.2,F4.2,2F3.1)
340 FJRMAT (1413)
350 FJRMAT (10F6.4/4F6.4)
360 FORMAT (14F4.2)
370 FJRMAT (8F6.4)
380 FJRMAT (8F6.4)
390 FJRMAT (10',7//'0',T30,'ALPHA= .1')
400 FJRMAT ('0',7//'0',T35,'RS=',F10.8,T55,'R(450)=',F10.8,T75,
1'RS-R(450)=',F10.8//'0',T35,'STANCARD DEVIATION =',
1F1C.8//'0',T35,'ACTUAL CONFIDENCE VALUE=',F6.2,'%')
420 FJRMAT ('0',7///'0',T30,'ALPHA= .2')
420 FJRMAT ('0',7///'0',T30,'ALPHA= .2')
420 FJRMAT ('0',7///'0',T35,'RS=',F10.8,T55,'RR(400)=',F10.8,T75
1,'RS-RR(4C0)=',F10.8//'0',T35,'STANDARD DEVIATION=',
1F10.8//'0',T35,'ACTUAL CONFIDENCE VALUE=',F6.2,'%')
END
```

BIBLIOGRAPHY

- Chow, T., Schiller, L. and Tomsky, J., "System Reliability Estimation from Several Data Sets," Proceedings 1976 Annual Reliability and Maintainability Symposium, pp. 18-24.
- Degroot, Morris H., <u>Probability and Statistics</u>, Addison-Wesley, 1975.
- Naval Postgraduate School Research Report NPS55LW73061A, Naval Postgraduate School Random Number Generator Package, by G.P. Learmonth and P.A.W. Lewis, June 1973.

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